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## FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL HARDWARE

NUMBER: 05-6N-2048 -X

SUBSYSTEM NAME: EPD&C - AUXILIARY POWER UNIT (04-2)

REVISION: 0

11/21/87

PART	DATA	

TRITURIA		
	PART NAME VENDOR NAME	PART NUMBER VENDOR NUMBER
LRU	: AFT LCA 1	MC450-D057-0001
LRU	: AFT LCA 2	MC450-D058-0001
LRU	; AFT LCA 3	MC450-0059-0001
SRU	: DIODE	JANTXV1N5551

# EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

DIODE, ISOLATION (3 AMP) - AUXILIARY POWER UNIT (APU) HEATERS, LUBE OIL LINE 1, 2, AND 3 POWER CIRCUITS

REFERENCE DESIGNATORS: 54V76A121(J3-15,J1-12)

> 54V76A121(J3-16,J1-12) 54V76A121(J3-31,J1-21) 54V76A121(J3-32,J1-21) 55V76A122(J3-15,J1-12) 55V76A122(J3-16,J1-12) 55V76A122(J3-31,J1-21) 55V76A122(J3-32,J1-21) 56V76A123(J3-15,J1-12) 56V76A123(J3-16,J1-12) 56V76A123(J3-31,J1-21) 56V76A123(J3-32,J1-21)

QUANTITY OF LIKE ITEMS: 12

TWELVE (4 PER APU)

## **FUNCTION:**

TO ISOLATE SWITCH SCAN CIRCUITS FROM THE HEATER CONTROL CIRCUITS OF LUBE OIL LINE 1, 2, AND 3.

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FAILURE MODES EFFECTS ANALYSIS FMEA - NON-CIL FAILURE MODE NUMBER: 05-6N-2048- 02

REVISION#:

01

08/01/96

ITEM NAME: DIODE

LRU: AFT LCA 1, 2, 3

SUBSYSTEM NAME: EPD&C - AUXILIARY POWER UNIT (04-2)

CRITICALITY OF THIS

FAILURE MODE: 183

FAILURE MODE:

SHORT (END TO END)

MISSION PHASE:

PL PRE-LAUNCH

LO LIFT-OFF

OO ON-ORBIT

DO DE-ORBIT

LS LANDING/SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY

104 **ATLANTIS** 

105 **ENDEAVOUR** 

CAUSE:

STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), CONTAMINATION. ELECTRICAL STRESS, THERMAL STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) N/A

C) PASS

PASS/FAIL RATIONALE:

Sp.

A)

DIODE IS 2-FAULT TOLERANT WITH 2 REMAINING LEGS VERIFIABLE IN FLIGHT. OPERATION OF THERMOSTAT IS VERIFIABLE IN FLIGHT, STATUS OF REMAINIG APUR IS. VERIFIABLE IN FLIGHT.

Ç)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

# FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL FAILURE MODE NUMBER: 05-6N-2046- 02

NO EFFECT - FIRST FAILURE

#### (B) INTERFACING SUBSYSTEM(S):

NO EFFECT - FIRST FAILURE

#### (C) MISSION:

NO EFFECT - FIRST FAILURE. ABORT DECISION REQUIRED AFTER THREE FAILURES - LOSS OF ONE APU.

#### (D) CREW, VEHICLE, AND ELEMENT(S):

NO EFFECT - FIRST FAILURE

#### (E) FUNCTIONAL CRITICALITY EFFECTS:

LOSS OF CREW/VEHICLE AFTER THREE OTHER FAILURES (SWITCH CONTACT FAILS CLOSED, THERMOSTAT CONTACTS FAIL CLOSED, LOSS OF SECOND APU) DUE TO LOSS OF TWO OF THREE APU'S.

#### -DISPOSITION RATIONALE-

#### (A) DESIGN:

REFER TO APPENDIX F, ITEM NO. 4 - DIODE

## (B) TEST:

REFER TO APPENDIX F, ITEM NO. 4 - DIODE

GROUND TURNAROUND TEST APU 1/2/3 LUBE OIL HEATER CIRCUIT TESTS PERFORMED EVERY FLOW.

#### (C) INSPECTION:

REFER TO APPENDIX F, ITEM NO. 4 - DIODE

## (D) FAILURE HISTORY:

REFER TO APPENDIX F, ITEM NO. 4 - DIODE

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( FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL FAILURE MODE NUMBER: 05-6N-2048- 02

(E) OPERATIONAL USE:

NONE

- APPROVALS -

EDITORIALLY APPROVED EDITORIALLY APPROVED

: RI : JSC

TECHNICAL APPROVAL

: VIA JSC

96-CIL-010

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